## Solving the Speed Run

How we broke through the 200kph barrier
Tijl Schmelzer \& Bert Sen. Schmelzer

## Preface

Just before Christmas 2014, I was preparing for our gliding trip to Kiripotib Flying Lodge in Namibia. I was updating the list of Belgian, Continental and World records with the newest record claims.

My eye had up till then been set on the "Koninginnenummer": the Open Class 3TP distance. My brother Bert and I had already broken this African record in 2012 in Morocco, but our claim was not ratified due to a silly bureaucratic reason. Because of our story, this bureaucratic process has since then been adjusted by the IGC/FAI, so our experience could not happen again.

However, in the last email of the IGC-mailing list, a new record claim for the free 3TP was just announced, with an impressive distance of 1349.4 km , flown by Bostjan Pristavec and Klaus Seemann. This record would be hard to beat. The good news for us though, is that we would be allowed to fly their glider in Namibia: the beautiful EB28 Edition "7".

In this IGC-email, the last claim caught my attention. Laszlo Hegedus had one day before broken Makoto Ichikawa's standing African record on the Speed over a triangular course of 100 km . He had improved the record from $171.83 \mathrm{~km} / \mathrm{h}$ to $192.69 \mathrm{~km} / \mathrm{h}$.

I wondered: "What would be the theoretical maximum for this speed run?"
Tijl

## 100km FAI triangle

The 100 km FAI triangle is the sprint discipline of the gliding athletics. It is very different from all other record categories: it is so simple, that you can optimize it so much that it can be completed almost perfectly according to the MacCready rules. Gliding tactics to deal with risk management are unimportant.

Additionally, all record categories allow for a max difference between start and finish altitude of 1000 m . This bonus results in a significant boost in the average speed, as it can be used for a final glide which doesn't have to be "earned" by thermaling. In the 100km triangle, this represents a relatively large part of the flight, and thus has a much larger impact than on the larger triangles and out-and-return categories.

So, the 100 km FAI triangle record can be defined as: One Perfect Climb, One Perfect Final Glide.
Its pureness makes it a very interesting theoretical and practical exercise.
However, it has to be said that a significant part of what constitutes gliding is excluded, and, as pilot's decision making skills play a minor role, you don't have to be a world champ to break this record. Much more important is planning, weather forecasting and task setting. And concentration during the execution, off course.

Above all, it's a lot of thrilling fun. The task only takes about half an hour, but this is time is very intense: you are incredibly focused, flying close to VNE, constantly looking at the clock and altimeter, and looking for the best line to get to the finishline as rapidly as possible. To me, it was almost as exciting as the final race of the World Gliding Championships in Rayskala last year.

## The rules

The task has to be a triangle with a startpoint, 2 turnpoints and a finishpoint. Since it has to be a closed course, the start and finishpoint have to be the same.

The startline and finishline both are lines of 1 km width ( 500 m radius), and the TP sectors are 90 degree sectors with 3 km radius.

Each leg has to be at least $28 \%$ of the total distance. This means that the largest leg is max $44 \%$ of the total distance.

And, finally, as mentioned above, the finish altitude cannot be lower than 1000 m under the start altitude.

## 100km Speed records

Some pilots, amongst whom Hans-Werner Grosse, have lobbied in the past to split up all record categories between thermal and dynamic lift records. It is indeed true that you can't fly as fast, far or high in thermal conditions as in wave. Both are pretty much incomparable, so it's useless to point at the fantastic World Record of $289.4 \mathrm{~km} / \mathrm{h}$ flown in the lee wave of the Andes by Klaus Ohlmann. This is far above the limit possible with thermals.

Luckily for us in this case, there hasn't yet been found a good wave playground in Africa. The 100km speed triangle performance lies thus still within the realm of the thermal flights. And in the last few years, there has been a bit of a rush to knock off seconds of this task.

Overview of African Continental 100km Triangle Progress (Open Class Category):

| Speed (km/h) | Time | Date | Pilot | Glider | Location |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 156.6 | 38:19 | 2012/05/31 | Laszlo Hegedus | Nimbus 4T | Bitterwasser,NAM |
| 169.8 | 35:20 | 2013/03/14 | Uys Joncker | JS1C-21m | Tswalu, SA |
| 171.83 | 34:55 | 2013/12/11 | Makoto Ichikawa | Ventus 2cxm | Bitterwasser,NAM |
| 175.58* | 34:10 | 2014/12/12 | Mauro Brunazzo | Quintus M | Bitterwasser,NAM |
| 192.95* | 31:08 | 2014/12/21 | Laszlo Hegedus | EB28 | Bitterwasser,NAM |
| 205.89* | 29:09 | 2014/01/06 | Tijl \& Bert Sen. Schmelzer | EB28 Edition | Kiripotib, NAM |
| *Claims - Not yet ratified |  |  |  | (Times are recalculated to 100km) |  |

## The 200kph barrier

A barrier as artificial as the 1000km mark. However, not a single gliding record in thermal conditions has been completed beyond this boundary, and most likely not a single set task neither. And thus, to me, it made this challenge even more exciting. This would entail finishing within 30 minutes of crossing the startline.

But first, we had to analyze if it was even possible to break this barrier.

## Some definitions

Before we start with the analysis, I would like to present you with a few definitions, which are common knowledge for pilots who are used to fly in areas with high cloudbases.

Indicated Air Speed (IAS): is the airspeed you read on your airspeed indicator.
True Air Speed (TAS): Since the air density and pressure decrease with altitude, less air molecules push on the membrane of the airspeed indicator. You are thus flying faster through the air, than airspeed indicates. True Air Speed compensates for this measurement error.

Groundspeed: True Air Speed + Windspeed
In higher altitudes the altitude effect becomes noticeable quickly: the difference between IAS and TAS increases with the square root of the air density.

$$
I A S=T A S \sqrt{\frac{\rho}{\rho 0}}
$$

With $\rho$ being the density of the air in which the glider is flying, and $\rho 0$ the air density in the International Standard Atmosphere $\left(15^{\circ} \mathrm{C}, 1013.25 \mathrm{hPa}=1.225 \mathrm{~kg} / \mathrm{m} 3\right)$.

Since temperature in Namibia is not very cold at cloudbase (mostly above freezing temperature), it is also reasonable to ignore temperature effects in our calculations. The above formula can thus be changed to a, for us easier to use, formula:

$$
I A S=T A S \sqrt{\frac{\text { Air Pressure }}{\text { Std.Pressure }}}
$$

With Std. Pressure being 1013.25 hPa
It is also reasonable to use the ICAO standard atmosphere (a standardized scale of how pressure changes with altitude) for calculations. An approximation for the ICAO standard atmosphere is given by the following formula:

$$
\text { Air Pressure }=(1 / 100)\left(\frac{(\text { Altitude }(m)-44330.8)}{4946.54}\right)^{1 / 0.1902632} h P a
$$

For example, using these formula's, an IAS of $200 \mathrm{~km} / \mathrm{h}$ at 5000 m ( 540 hPa ), would be equal to a TAS of $273.96 \mathrm{~km} / \mathrm{h}$.

The importance of these definitions will be shown later on.

## Quick analysis of Laszlo Hegedus' Record Claim flight

So, this analysis started with taking a closer look at the best performance up till then. Laszlo Hegedus is a member of the Hungarian National team, and has competed in many WGC's. He holds the World Record of the Speed over triangular course of 1250 km at $151.1 \mathrm{~km} / \mathrm{h}$. (Bitterwasser, NAM on Nimbus 4t) This is currently one of only two open class world record NOT flown in wave!

During his 100km record flight, he made two attempts, and in the final one he flew a speed of $192.95 \mathrm{~km} / \mathrm{h}$.
The vital statistics on this run, with my short comments:

| Statistic | Value | Unit | Comment |
| :--- | ---: | :--- | :--- |
| Task Length | 119.3 | km | Why longer than 100km? |
| Legs | $32.0 \%-37.6 \%-30.4 \%$ |  | Doesn't look optimal. |
| Cloudbase | 5073 | m |  |
| Start Alt. | 4875 | m |  |
| Finish Alt. | 4078 | m |  |
| Alt. Difference | 797 | m | 203 meter left over. |
| \# of Climbs | 1 |  | Good |
| Climb Rate | 4.8 | $\mathrm{~m} / \mathrm{s}$ | Very powerful! |
| Alt. Gain | 600 | m | Low |
| Time In Climb | $02: 06$ |  | Very low |
| Total Distance | 125.1 | km |  |
| Detours | 5.8 | km |  |
| Detours | $4.86 \%$ |  | Low for normal flight, rather average high for |
| Mean L/D | 88 |  | 100km speed |
| Total Avg. | 215 | $\mathrm{~km} / \mathrm{h}$ | Too low |
| Groundspeed |  |  |  |
| Total Avg. IAS | 172 | $\mathrm{~km} / \mathrm{h}$ | Too low |
| Rising Avg. GS | 184 | $\mathrm{~km} / \mathrm{h}$ | About right |
| Rising Avg. IAS | 151 | $\mathrm{~km} / \mathrm{h}$ | About right |
| Sinking Avg. GS | 232 | $\mathrm{~km} / \mathrm{h}$ | Way too low |
| Sinking Avg. IAS | 183 | $\mathrm{~km} / \mathrm{h}$ | Way too low |
|  |  |  |  |

My main conclusions from this flight, is that it is very good, but not perfect. The climb rate and mean glide ratio are extremely high. However, the glidespeed in sinking air is too low speed (as I will show in the MacCready calculations). Furthermore, the finish was too high.

So there is still room for improvement.

## Optimization

## The EB28 Edition

All gliding optimization calculations start with the polar of the glider at hand.
The quadratic approximated sink polar is defined as:

$$
w=a V^{2}+b V+c
$$

With $V$ the airspeed ( $\mathrm{km} / \mathrm{h}$ ), and $w$ the corresponding sinkspeed ( $\mathrm{m} / \mathrm{s}$ ). By dividing airspeed V by sinkspeed w , we get the glide (LD) polar.

The $a, b$ and $c$ coefficients are inherent to each glider and wingloading, and the one used for the EB28 edition at min. weight are $a=-0.00011616, b=0.02182417, c=-1.4585725$

Note that $a, b$ and $c$ values are different from your flight computer or analysis software values, because of different units used in these formulas. They still represent the same polar.


This sink and glide ratio polar seem to be a bit optimistic at low speeds (max LD 65.5), however, I think it is realistic in the speed range that matters for this flight ( $150 \mathrm{~km} / \mathrm{h}-250 \mathrm{~km} / \mathrm{h}$ IAS).

## The effect of waterballast

Off course, weight of the glider affects the polar. If we increase or decrease the weight, for instance by adding water ballast, the polar will change. Because of the underlying aerodynamical physics, this change can be very well approximated by scaling the original sink polar around the origin.


The scaling factor W is:

$$
W=\sqrt{\frac{\text { Weight New }}{\text { Weight Original }}}
$$

For the $\mathrm{a}, \mathrm{b}$ and c coefficients of the sinkpolar the effect is:

$$
\begin{gathered}
a^{\prime}=\frac{a}{W} \\
b^{\prime}=b \\
c^{\prime}=W c
\end{gathered}
$$

If you inquire the effects of this formula, you find that a weight increase does not affect the glide ratio of a glider, it just increases the airspeed at which the best glide ratio can be found. In reality, a weight increase does in fact increase the glide ratio slightly due to flying at higher Reynolds numbers. This, rather small, effect is thus neglected in all glide computers and optimization programs.

Off course, since minimum sink will be higher (and at a higher speed), climbrates in thermals will suffer a bit.

For the 100km speed run, which can only be flown in fantastic conditions, a glider cannot be heavy enough. We will thus need to fly the EB28 at 850 kg , with a wingloading of $51.5 \mathrm{~kg} / \mathrm{m} 2$. The resulting polar can be found in the following figure.


## The altitude effect

Altitude not only has a tremendous effect on the IAS measurement, but also on the real sinkpolar and glide polar.

At higher altitude, less air molecules are available to carry the plane. This means the sink speed of the glider will increase with altitude.

But also, there are less air molecules to hold the glider back. Drag is thus reduced as well, making it easier to fly faster.

The net impact of both sink and forward speed can be expressed exactly in the same way as adding waterballast to the glider: it is a scaling of the sink polar around the origin.


The scaling factor $P$ you are already familiar with:

$$
P=\sqrt{\frac{\text { Air Pressure }}{\text { Std. Pressure }}}
$$

This again ignores secondary temperature effects. And the formula also ignores the small increase in glide ratio in altitude due to the higher Reynolds numbers.

So this altitude effect, a part from the great climb rates, is one of the main reasons people fly so fast in Namibia. As a comparison: the polar of an EB28 at 5000 m at $51.5 \mathrm{~kg} / \mathrm{m} 2$ ( 850 kg ) corresponds to an EB28 at sea level at $97 \mathrm{~kg} / \mathrm{m} 2(1600 \mathrm{~kg})$ !


Something else that is important related to the altitude effect: How does the redline change with altitude?
This is an engineering question, which is much more difficult to answer in general. It has to do with the damping effects on flutter of the airmass and many other factors. Normally, the manufacturer of the glider includes a table with the change of max IAS with altitude. For the EB28 Edition, the max IAS decreases from $285 \mathrm{~km} / \mathrm{h}$ at sea level to $245 \mathrm{~km} / \mathrm{h}$ at 5000 m . This corresponds with an increase of TAS to $335.6 \mathrm{~km} / \mathrm{h}$ !

For the yellow line, such a table is not commonly available, so common sense prevails. I would not fly at IAS $245 \mathrm{~km} / \mathrm{h}$ in 5000 m trough a $5 \mathrm{~m} / \mathrm{s}$ thermal.

So, the question becomes: How does the altitude effect affect MacCready rules and Speed To Fly?
You know that, when you increase waterballast, your Speed To Fly for a certain MacCready Value increases as well. That's why you have to always put in the right wingloading into your flightcomputer.

Since the altitude effect on the polar is similar to the addition of waterballast, you might deduce that you should also fly faster in higher altitude for the same MC setting. And in fact, such is the case. However, we are talking in terms TAS, not IAS!

A following explanation can be often heard or even found in textbooks: "Quite neatly, the altitude effect on Speed To Fly, and the altitude effect on the difference between IAS and TAS cancel each other out. So we can use the IAS measurement for the Speed To Fly as calculated for sea level. This makes it easy, so we just can keep our STF steady looking at the airspeed indicator, while descending during a glide."

However, this is not true! The following tables shows the optimal speed to fly in IAS and TAS in 4 altitudes ( $0 \mathrm{~m}, 3000 \mathrm{~m}, 4000 \mathrm{~m}, 5000 \mathrm{~m}$ ) for a MacCready Value of $4.5 \mathrm{~m} / \mathrm{s}$, in still air, and with vertical airmass movement $(+3 \mathrm{~m} / \mathrm{s},+2 \mathrm{~m} / \mathrm{s},+1 \mathrm{~m} / \mathrm{s},+0.5 \mathrm{~m} / \mathrm{s}, 0 \mathrm{~m} / \mathrm{s},-0.5 \mathrm{~m} / \mathrm{s},-1 \mathrm{~m} / \mathrm{s},-2 \mathrm{~m} / \mathrm{s},-3 \mathrm{~m} / \mathrm{s})$ :

## Optimal Speed To Fly EB28Edition@51.5kg/m2

| IAS | MC4.5 |  |  |  |
| :---: | :---: | :---: | :---: | :---: |
|  | Om | 3000m | 4000m | 5000m |
| 3.0 | 167 | 160 | 158 | 156 |
| 2.0 | 193 | 182 | 179 | 176 |
| 1.0 | 215 | 202 | 198 | 194 |
| 0.5 | 225 | 211 | 207 | 202 |
| Still Air | 235 | 220 | 215 | 210 |
| -0.5 | 245 | 229 | 223 | 218 |
| -1.0 | 254 | 237 | 231 | 226 |
| -2.0 | 272 | 252 | 246 | 240 |
| -3.0 | 288 | 267 | 260 | 254 |

TAS MC4.5

| Om | $\mathbf{3 0 0 0 m}$ |  | 4000m | 5000m |
| :---: | :---: | :---: | :---: | :---: | :---: |
| $\mathbf{3 . 0}$ | 167 | 192 | 202 | 213 |
| $\mathbf{2 . 0}$ | 193 | 219 | 230 | 241 |
| $\mathbf{1 . 0}$ | 215 | 243 | 254 | 266 |
| $\mathbf{0 . 5}$ | 225 | 254 | 265 | 277 |
| Still Air | 235 | 265 | 276 | 288 |
| $\mathbf{- 0 . 5}$ | 245 | 275 | 286 | 299 |
| $\mathbf{- 1 . 0}$ | 254 | 285 | 297 | 309 |
| $\mathbf{- 2 . 0}$ | 272 | 303 | 316 | 329 |
| $\mathbf{- 3 . 0}$ | 288 | 321 | 334 | 347 |

Although modern flight computers could in theory incorporate this, most if not all currently still ignore this secondary effect, as it is not really that important for the vast majority of our performances. In the case of the perfect 100 km speed run, it is.

From the table you can see, that STF based on IAS decreases with altitude. Based on TAS it, off course, increases with altitude.

Interestingly, in 5000m with MC4.5, even if you fly through a decent thermal of $2.0 \mathrm{~m} / \mathrm{s}$ during your glide, while your IAS should not go below $176 \mathrm{~km} / \mathrm{h}$, your TAS should not go below $241 \mathrm{~km} / \mathrm{h}$ ! This means that you have to keep your speed up, even while crossing decent climbs. In reality, this is even more pronounced, since dynamic transition losses are not accounted for in these formulas. Thus: don't pull too much while crossing climbs.

Additionally, there is also the issue that the same IAS-TAS difference because of altitude can also be found in certain variometer types. Some vario's show not the real climb and sink rates. And this has an effect on correct MC Value setting. I'd like to refer to Reichmann's Streckensegelflug for more information on that. Most modern common electric vario's do measure true climb and sink rates.

How does this all translate into cross country average speeds?
If we look at the case where there is no vertical airmass movement, the following tables give the summation for Sea level and 4000 m :

| Sea level | Climbrate | 3 | 3.5 | 4 | 4.5 | 5 | 5.5 | $6 \mathrm{~m} / \mathrm{s}$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | STF TAS = IAS | 204 | 215 | 225 | 235 | 245 | 254 | 263 km/h |
|  | Sinkspeed True | -1.65 | -1.91 | -2.18 | -2.47 | -2.76 | -3.06 | -3.36 m/s |
|  | LD | 34.4 | 31.3 | 28.7 | 26.5 | 24.7 | 23.1 | 21.7 |
| True XC | Alt.Loss | -2905 | -3195 | -3484 | -3770 | -4053 | -4331 | -4605 m |
|  | TimeToClimb | 16.1 | 15.2 | 14.5 | 14.0 | 13.5 | 13.1 | 12.8 min |
|  | TimeToCruise | 29.4 | 27.9 | 26.6 | 25.5 | 24.5 | 23.6 | 22.8 min |
|  | TotalTime | 45.5 | 43.1 | 41.1 | 39.5 | 38.0 | 36.7 | 35.6 min |
|  | XCSpeed | 131.8 | 139.2 | 145.9 | 152.1 | 157.9 | 163.3 | 168.5 km/h |
| With 1000m diff. | Alt.Loss | -1905 | -2195 | -2484 | -2770 | -3053 | -3331 | -3605 m |
| Start-Finish | TimeToClimb | 10.6 | 10.5 | 10.4 | 10.3 | 10.2 | 10.1 | 10.0 min |
|  | TotalTime | 40.0 | 38.4 | 37.0 | 35.7 | 34.7 | 33.7 | 32.8 min |
|  | Avg. Speed | 150.1 | 156.5 | 162.3 | 167.8 | 173.1 | 178.0 | 182.8 km/h |


| 4000m | Climbrate | 3 | 3.5 | 4 | 4.5 | 5 | 5.5 | $6 \mathrm{~m} / \mathrm{s}$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | STF TAS | 242 | 254 | 265 | 276 | 286 | 297 | 306 km/h |
|  | STF IAS | 189 | 198 | 207 | 215 | 223 | 231 | $239 \mathrm{~km} / \mathrm{h}$ |
|  | Sinkspeed True | -1.70 | -1.94 | -2.19 | -2.45 | -2.73 | -3.01 | -3.29 m/s |
|  | LD | 39.6 | 36.4 | 33.6 | 31.2 | 29.2 | 27.4 | 25.8 |
| True XC | Alt.Loss | -2523 | -2748 | -2975 | -3201 | -3427 | -3650 | -3872 m |
|  | TimeToClimb | 14.0 | 13.1 | 12.4 | 11.9 | 11.4 | 11.1 | 10.8 min |
|  | TimeToCruise | 24.8 | 23.6 | 22.6 | 21.7 | 20.9 | 20.2 | 19.6 min |
|  | TotalTime | 38.8 | 36.7 | 35.0 | 33.6 | 32.4 | 31.3 | 30.3 min |
|  | xCSpeed | 154.6 | 163.4 | 171.3 | 178.6 | 185.4 | 191.7 | 197.7 km/h |
| With 1000m diff. | Alt.Loss | -1523 | -1748 | -1975 | -2201 | -2427 | -2650 | -2872 m |
| Start-Finish | TimeToClimb | 8.5 | 8.3 | 8.2 | 8.2 | 8.1 | 8.0 | 8.0 min |
|  | TotalTime | 33.3 | 32.0 | 30.9 | 29.9 | 29.0 | 28.3 | 27.6 min |
|  | Avg. Speed | 180.4 | 187.7 | 194.5 | 200.7 | 206.7 | 212.3 | 217.6 km/h |

Now this is uplifting news! According to this, with a realistic $4.5 \mathrm{~m} / \mathrm{s}$ climb rate, and an average altitude of 4000 m during the flight, an average speed of $200.7 \mathrm{~km} / \mathrm{h}$ should be possible. Even without cloudstreeting and energy lines!

Also, notice that the average speed bonus of the 1000 m altitude difference between start and finish is about $20 \mathrm{~km} / \mathrm{h}$. So, people thinking the 100 km triangle is just a simple final glide, and nothing to it, are wrong. Beating the $200 \mathrm{~km} / \mathrm{h}$ barrier, still means flying above $180 \mathrm{~km} / \mathrm{h}$ true average XC speed, albeit over a relatively short distance!

Unfortunately, some things are not taken into account:

- Detours: in a normal flight, you fly $5-10 \%$ more km than the task distance because the optimal line is very rarely straight on course. In the 100 km speed run, each $\%$ detour, results in a bit more than a $2 \mathrm{~km} / \mathrm{h}$ speed loss. A realistic value lies between $2.5-5 \%$, so $5-10 \mathrm{~km} / \mathrm{h}$ speed loss.
- You lose speed while cornering around the turning points. First you have to slow down and speed up again, which results in dynamic losses as well as deviating from optimal STF. And, secondly, you have to fly a bit (about 250 m ) beyond the turnpoint. To be certain to have a log fix within the turnpoint sector. I estimate the losses to be ca $1.5 \mathrm{~km} / \mathrm{h}$ per turnpoint, so $3 \mathrm{~km} / \mathrm{h}$ in total. This part can be included in the detours.
- The pilot, nor the glider is perfect. At these high speeds, small imperfections (deviations from optimal STF, time to center climb, slipping, not using total 1000 m buffer,...), cause significant losses in average speed. It is hard to estimate exactly how much, but I think it should be in the range of $5-10 \mathrm{~km} / \mathrm{h}$.

Adding these three losses together we come to a realistic max speed on the 100 km triangle of $13-25 \mathrm{~km} / \mathrm{h}$.
This reduces the attainable record speed to $175-188 \mathrm{~km} / \mathrm{h}$ with a realistic $4.5 \mathrm{~m} / \mathrm{s}$ climb, and $187-199 \mathrm{~km} / \mathrm{h}$ with a fantastic $5.5 \mathrm{~m} / \mathrm{s}$ climb.

Luckily, there are some pieces of the puzzle left in the bag.

## Dynamic pull

There is no max limit to the startspeed, and no min limit to the finishspeed. You could thus start at VNE, and finish with stall speed. In a normal flight, this is quite unimportant, but, in the 100 km task, it can be vital!

In a real situation, you would start at 4500 m with IAS $235 \mathrm{~km} / \mathrm{h}$ (TAS $311 \mathrm{~km} / \mathrm{h}$ ), and finish in 3500 m with IAS $110 \mathrm{~km} / \mathrm{h}$ (TAS $137 \mathrm{~km} / \mathrm{h}$ ). You would off course pull up a bit from IAS $235 \mathrm{~km} / \mathrm{h}$ to ca IAS $215 \mathrm{~km} / \mathrm{h}$ just after the start, and from IAS $215 \mathrm{~km} / \mathrm{h}$ to IAS $100 \mathrm{~km} / \mathrm{h}$ just before the finish. But for simplicity, it is easier to see it as one pull-up.

How does speed convert to altitude? Ignoring second order air resistance effects, the physics are quite simple by equating kinetic and potential energy:

$$
E \frac{m V^{2}}{2}=m g h
$$

With m mass, V True Air Speed in $\mathrm{m} / \mathrm{s}$, dimensionless E dynamic transition efficiency, g gravitational constant $9.8 \mathrm{~m} 3 /(\mathrm{kg} . \mathrm{s} 2)$, and h altitude in m .

Since the efficiency of a single transition in a glider is rather efficient, we can arbitrarily put this at $90 \%$. This value is likely a bit too conservative.

Since the mass can be scrapped from both sides, this becomes:

$$
\Delta h=E \frac{V 0^{2}-V^{2}}{2 g}
$$

with V0 the original speed in $\mathrm{m} / \mathrm{s}$, and $\Delta h$ the change in altitude.
Filling in the above speeds of $311 \mathrm{~km} / \mathrm{h}$ and $137 \mathrm{~km} / \mathrm{h}$, gives us a $\Delta h$ of 300 m .
Off course, it is impossible to do and time this maneuver perfectly, so let's round it down to 200 m . The following table shows the effect on the average speed:

| 4000m | Climbrate | 3 | 3.5 | 4 | 4.5 | 5 | 5.5 | $6 \mathrm{~m} / \mathrm{s}$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | STF TAS | 242 | 254 | 265 | 276 | 286 | 297 | 306 km/h |
|  | STF IAS | 189 | 198 | 207 | 215 | 223 | 231 | $239 \mathrm{~km} / \mathrm{h}$ |
|  | Sinkspeed True | -1.70 | -1.94 | -2.19 | -2.45 | -2.73 | -3.01 | -3.29 m/s |
|  | LD | 39.6 | 36.4 | 33.6 | 31.2 | 29.2 | 27.4 | 25.8 |
| True XC | Alt.Loss | -2523 | -2748 | -2975 | -3201 | -3427 | -3650 | -3872 m |
|  | TimeToClimb | 14.0 | 13.1 | 12.4 | 11.9 | 11.4 | 11.1 | 10.8 min |
|  | TimeToCruise | 24.8 | 23.6 | 22.6 | 21.7 | 20.9 | 20.2 | 19.6 min |
|  | TotalTime | 38.8 | 36.7 | 35.0 | 33.6 | 32.4 | 31.3 | 30.3 min |
|  | xCSpeed | 154.6 | 163.4 | 171.3 | 178.6 | 185.4 | 191.7 | 197.7 km/h |
| With 1000 m diff. | Alt.Loss | -1523 | -1748 | -1975 | -2201 | -2427 | -2650 | -2872 m |
| Start-Finish | TimeToClimb | 8.5 | 8.3 | 8.2 | 8.2 | 8.1 | 8.0 | 8.0 min |
|  | TotalTime | 33.3 | 32.0 | 30.9 | 29.9 | 29.0 | 28.3 | 27.6 min |
|  | Avg. Speed | 180.4 | 187.7 | 194.5 | 200.7 | 206.7 | 212.3 | 217.6 km/h |
| With 1000 m diff. | Alt.Loss | -1323 | -1548 | -1775 | -2001 | -2227 | -2450 | -2672 m |
| Start-Finish | TimeToClimb | 7.4 | 7.4 | 7.4 | 7.4 | 7.4 | 7.4 | 7.4 min |
| and Pull-Up | TotalTime | 32.1 | 31.0 | 30.0 | 29.1 | 28.4 | 27.7 | 27.0 min |
|  | Avg. Speed | 186.7 | 193.5 | 199.9 | 205.8 | 211.5 | 216.9 | 222.1 km/h |

The Pull-up gives us thus an additional $5 \mathrm{~km} / \mathrm{h}$ bonus. We are now nearing the possibility of the $200 \mathrm{~km} / \mathrm{h}$ barrier, but only with a $5.5 \mathrm{~m} / \mathrm{s}$ climb.

## Cloudstreeting

Luckily, there are cloudstreets and energylines. This will be the final piece of the puzzle to the 100 km record. But also the hardest to put into a model.

Energylines shift the sink polar upwards, with its upward velocity. This means that your glide ratio will improve tremendously. The longer, and the better the patches of rising air, the more we can boost our average speed. So, what does a realistic Cloudstreet look like?

For this we would need a general distribution of vertical airmass movement along the track. My preferred method for this, would be using actual flight data of a good flight. However, to extract the airmass movement from an IGC-file, a dynamic model would be needed, and I am not aware of such software currently existing.

Generally, I would split up the data in rising and sinking parts, and regard both of these parts as two separate homogenous blocks. The parameters could then be extracted from SeeYou.

However, SeeYou's paramaters are not derived from a dynamic model, and as such, netto climb and sink rates are exaggerated due to "Knuppeltermik". In moderate European circumstances on a $300 \mathrm{~km}+\mathrm{flight}$, this is not a big issue. For the purposes presented here, the skew in the data becomes very large.

In this case, I thus prefer to see the whole airmass as one homogenous entity. The vertical movement is then derived from comparing the actual average sinkrate during the total glide, to the sinkrate connected to the actual average glidespeed as derived from the theoretical EB28 polar in the average altitude.

In this way, the effect of dynamic transitions is eliminated. The model thus becomes a bit less accurate, but more robust. In reality, one would fly slower than the model STF in rising parts, and faster than the STF in sinking parts. In reality, the glide angle, and the final average task speed would both be a bit (a few $\mathrm{km} / \mathrm{h}$ ) higher. This is important keep in mind looking at the following pages.

To find realistic values, we can use the values of Laszlo Hegedus' record flight:

| Alt. Loss in Glide | -1414 m |
| :--- | :---: |
| Time Glide | 1982 sec |
| Average Sinkrate | $-0.71 \mathrm{~m} / \mathrm{s}$ |
| Average Glidespeed TAS | $215 \mathrm{~km} / \mathrm{h}$ |
| Polar Sinkrate for that TAS | $-1.17 \mathrm{~m} / \mathrm{s}$ |
| Netto Avg. Airmass movement | $0.46 \mathrm{~m} / \mathrm{s}$ |

## Analysis and optimization of the 192.95 km Record Claim

## The actual performance

Putting all data in the simple model delivers a resulting time ( $37: 20$ ) and speed of $191.7 \mathrm{~km} / \mathrm{h}$ very close to the actual record claim ( $37: 05$ ) and $192.95 \mathrm{~km} / \mathrm{h}$.

From previous experience with using this simple model, this accuracy is not uncommon, since it is fitted to the polar.

Also the glide ratio from the model (84) and reality (88) are very close, as well as the climb height in thermals ( 691 m vs 600 m ).

So, now that the model works, and since we have a real life scenario, we can start optimizing.

| Taskparameters |  |  |
| :--- | :--- | :--- |
|  | Taskdistance | 119.3 km |
|  | Total Distance | 125.1 km |
|  | De-tours | $4.86 \%$ |
|  |  | 4500 m |
|  | Average Altitude | 1.32 |
|  | Altitude Effect Factor | 803 m |
|  | Start-Finish Alt Difference | $4.76 \mathrm{~m} / \mathrm{s}$ |
|  | Climbrate |  |
|  |  | $0.46 \mathrm{~m} / \mathrm{s}$ |
|  | Netto cloud streeting climb |  |


| Output |  |  |
| :--- | :--- | :--- |
|  | timeclimb | 2.4 min |
|  | Altclimb | 691 m |
|  | timeglide | 34.9 min |
|  | LD | 84 |
|  |  |  |
|  | Total time | $\mathbf{3 7 : 2 0} \mathbf{~ m i n : s e c}$ |
|  | Average Speed |  |


| Calculations |  |  |
| :--- | :--- | :---: |
| Thermalling | Average netto climb | $5.56 \mathrm{~m} / \mathrm{s}$ |
| Glide | GlideSpeed TAS | $215 \mathrm{~km} / \mathrm{h}$ |
|  | GlideSpeed IAS | $162 \mathrm{~km} / \mathrm{h}$ |
|  | Sinkspeed True | $-0.71 \mathrm{~m} / \mathrm{s}$ |
|  | Time Glide | 34.9 min |
|  | Altitude loss | -1494 m |
|  | Time Climb | 5.2 min |
|  |  |  |
| Finalglide | Time not climb | -2.8 min |
|  |  |  |

## Flying the perfect MacCready Speed To Fly

The first, and most important part is the perfection of STF. The main improvement would be changing the average glidespeed from $215 \mathrm{~km} / \mathrm{h}$ TAS to the optimal STF of $278 \mathrm{~km} / \mathrm{h}$ TAS ( $210 \mathrm{~km} / \mathrm{h}$ IAS). The result of doing only this, would lead to an average speed of $204.8 \mathrm{~km} / \mathrm{h}$ !

However, glide ratio decreases to 41 , and required altitude to climb to 2264 m , which could not be done using a single thermal (see discussion between 1 and multiple thermals later on)!

Keep in mind, that this a hindsight optimization. And hindsight is 20/20.

| Taskparameters |  |  |
| :--- | :--- | :--- |
|  | Taskdistance | 119.3 km |
|  | Total Distance | 125.1 km |
|  | De-tours | $4.86 \%$ |
|  |  | 4500 m |
|  | Average Altitude | 1.32 |
|  | Altitude Effect Factor | 803 m |
|  | Start-Finish Alt Difference | $4.76 \mathrm{~m} / \mathrm{s}$ |
|  | Climbrate | $0.46 \mathrm{~m} / \mathrm{s}$ |
|  |  |  |
|  |  |  |


| Output |  |  |
| :--- | :--- | :---: |
|  | timeclimb | 7.9 min |
|  | Altclimb | 2264 m |
|  | timeglide | 27.0 min |
|  | LD | 41 |
|  |  |  |
|  | Total time | $\mathbf{3 4 : 5 7} \mathbf{~ m i n : s e c}$ |
|  | Average Speed |  |
|  |  |  |


| Calculations |  |  |
| :--- | :--- | :---: |
| Thermalling | Average netto climb | $5.56 \mathrm{~m} / \mathrm{s}$ |
|  |  |  |
|  | Glide | GlideSpeed TAS |
|  | Sinkspeed IAS | $278 \mathrm{~km} / \mathrm{h}$ |
|  | Time Glide | $210 \mathrm{~km} / \mathrm{h}$ |
|  | Altitude loss | $-1.89 \mathrm{~m} / \mathrm{s}$ |
|  | Time Climb | 27.0 min |
|  |  | -3067 m |
|  | Time not climb | 10.7 min |
|  |  | -2.8 min |

## Watching the red line

If we use a bit more conservative glidespeed of TAS $260 \mathrm{~km} / \mathrm{h}$ (IAS $196 \mathrm{~km} / \mathrm{h}$ ), and glidespeed), the average speed would still have been $203.9 \mathrm{~km} / \mathrm{h}$, but Glide ratio would be a healthy 48 , and required altitude to climb in the thermal 1738 m .

| Taskparameters |  |  |
| :--- | :--- | :--- |
|  | Taskdistance | 119.3 km |
|  | Total Distance | 125.1 km |
|  | De-tours | $4.86 \%$ |
|  |  |  |
|  | Average Altitude | 4500 m |
| Altitude Effect Factor | 1.32 |  |
|  | Start-Finish Alt Difference | 803 m |
| Climbrate | $4.76 \mathrm{~m} / \mathrm{s}$ |  |
|  |  |  |
|  | Netto cloud streeting climb | $0.46 \mathrm{~m} / \mathrm{s}$ |


| Output |  |  |
| :--- | :--- | :---: |
|  | timeclimb | 6.2 min |
|  | Altclimb | 1783 m |
|  | timeglide | 28.9 min |
|  | LD | 48 |
|  |  |  |
|  | Total time | $\mathbf{3 5 : 0 7} \mathbf{~ m i n : s e c}$ |
|  | Average Speed | $\mathbf{2 0 3 . 9} \mathbf{~ k m} / \mathbf{h}$ |


| Calculations |  |  |
| :--- | :--- | :---: |
| Thermalling | Average netto climb | $5.56 \mathrm{~m} / \mathrm{s}$ |
| Glide | GlideSpeed TAS | $260 \mathrm{~km} / \mathrm{h}$ |
|  | GlideSpeed IAS | $196 \mathrm{~km} / \mathrm{h}$ |
|  | Sinkspeed True | $-1.49 \mathrm{~m} / \mathrm{s}$ |
|  | Time Glide | 28.9 min |
|  | Altitude loss | -2586 m |
|  | Time Climb | 9.1 min |
|  | Time not climb | -2.8 min |
| Finalglide |  |  |

## Using the whole 1000m buffer and Pulling-Up before the finishline

Now, if we optimize this further, by using up 950 m of the 1000 m altitude difference, instead of the 803 m used, and additionally adding the 200 m of the pull-up before the finishline, we can increase the average speed further to $211.2 \mathrm{~km} / \mathrm{h}$.

| Taskparameters |  |  |
| :--- | :--- | :--- |
|  | Taskdistance | 119.3 km |
|  | Total Distance | 125.1 km |
|  | De-tours | $4.86 \%$ |
|  |  | 4500 m |
|  | Average Altitude | 1.32 |
|  | Altitude Effect Factor | 1150 m |
|  | Start-Finish Alt Difference | $4.76 \mathrm{~m} / \mathrm{s}$ |
|  | Climbrate |  |
|  |  |  |
|  | Netto cloud streeting climb |  |
|  |  |  |


| Output |  |  |
| :--- | :--- | :---: |
|  | timeclimb | 5.0 min |
|  | Altclimb | 1436 m |
|  | timeglide | 28.9 min |
|  | LD | 48 |
|  |  |  |
|  | Total time | $\mathbf{3 3 : 5 4} \mathbf{~ m i n : s e c}$ |
|  | Average Speed | $\mathbf{2 1 1 . 2 ~ \mathbf { ~ k m } / \mathbf { h }}$ |


| Calculations |  |  |
| :--- | :--- | :---: |
| Thermalling | Average netto climb | $5.56 \mathrm{~m} / \mathrm{s}$ |
|  |  |  |
|  | Glide | GlideSpeed TAS |
|  | Sinkspeed IAS True | $260 \mathrm{~km} / \mathrm{h}$ |
|  | Time Glide | $196 \mathrm{~km} / \mathrm{h}$ |
|  | Altitude loss | $-1.49 \mathrm{~m} / \mathrm{s}$ |
|  | Time Climb | 28.9 min |
|  |  | -2586 m |
|  | Time not climb | 9.1 min |
| Finalglide |  | -4.0 min |

## Why fly more than necessary?

A next step, is reducing the task distance from 119.3 km to 101 km . This improves the average speed already to $215.9 \mathrm{~km} / \mathrm{h}$ !

And now, the altitude needed to climb is reduced to a very manageable 1038 m , possible to do in a single climb!

This flight would thus have been possible in these circumstances (however, we would have needed to start 300 meter lower, but, this discussion will follow later).

| Taskparameters |  |  |
| :--- | :--- | ---: |
|  | Taskdistance | 101 km |
|  | Total Distance | 105.9 km |
|  | De-tours | $4.83 \%$ |
|  |  |  |
|  | Average Altitude | 4500 m |
|  | Altitude Effect Factor | 1.32 |
|  | Start-Finish Alt Difference | 1150 m |
|  | Climbrate | $4.76 \mathrm{~m} / \mathrm{s}$ |
|  |  |  |
|  | Netto cloud streeting climb | $0.46 \mathrm{~m} / \mathrm{s}$ |


| Output |  |  |
| :--- | :--- | :---: |
|  | timeclimb | 3.6 min |
|  | Altclimb | 1038 m |
|  | timeglide | 24.4 min |
|  | LD | 48 |
|  |  |  |
|  | Total time | $\mathbf{2 8 : 0 4} \mathbf{~ m i n : s e c}$ |
|  | Average Speed | $\mathbf{2 1 5 . 9} \mathbf{~ k m} / \mathbf{h}$ |


| Calculations |  |  |
| :--- | :--- | :---: |
| Thermalling | Average netto climb | $5.56 \mathrm{~m} / \mathrm{s}$ |
| Glide | GlideSpeed TAS | $260 \mathrm{~km} / \mathrm{h}$ |
|  | GlideSpeed IAS | $196 \mathrm{~km} / \mathrm{h}$ |
|  | Sinkspeed True | $-1.49 \mathrm{~m} / \mathrm{s}$ |
|  | Time Glide | 24.4 min |
|  | Altitude loss | -2188 m |
|  | Time Climb | 7.7 min |
|  | Time not climb | -4.0 min |

## Perfect weather

The $192.95 \mathrm{~km} / \mathrm{h}$ record claim was flown in very good, but not perfect weather. What would be the ultimate weather for the 100 km speed triangle?

This question is off course impossible to answer 100\% correctly, but we can make a good realistic guess, by increasing the Climbrate from $4.76 \mathrm{~m} / \mathrm{s}$ to a fantastic but possible climb of $5.5 \mathrm{~m} / \mathrm{s}$, and additionally increasing the Netto cloud streeting climb rate from 0.46 to $0.75 \mathrm{~m} / \mathrm{s}$.

At the same time, we decrease the detours to $3.5 \%$.
In this case, the max attainable average speed increases to $236.3 \mathrm{~km} / \mathrm{h}$, and I think this should be pretty close to the theoretical max record speed in thermal conditions on the EB28 Edition.

However, this would be extremely difficult to do in reality!

| Taskparameters |  |  |
| :--- | :--- | ---: |
|  | Taskdistance | 101 km |
|  | Total Distance | 104.5 km |
|  | De-tours | $3.50 \%$ |
|  |  |  |
|  | Average Altitude | 4500 m |
|  | Altitude Effect Factor | 1.32 |
|  | Start-Finish Alt Difference | 1150 m |
|  | Climbrate | $5.50 \mathrm{~m} / \mathrm{s}$ |
|  | Netto cloud streeting climb | $0.75 \mathrm{~m} / \mathrm{s}$ |
|  |  |  |


| Output |  |  |
| :--- | :--- | :---: |
|  | timeclimb | 3.8 min |
|  | Altclimb | 1260 m |
|  | timeglide | 21.8 min |
|  | LD | 43 |
|  |  |  |
|  | Total time | $\mathbf{2 5 : 3 8} \mathbf{~ m i n : s e c}$ |
|  | Average Speed | $\mathbf{2 3 6 . 3} \mathbf{~ k m} / \mathbf{h}$ |


| Calculations |  |  |
| :--- | :--- | ---: |
| Thermalling | Average netto climb | $6.30 \mathrm{~m} / \mathrm{s}$ |
|  | Glide | GlideSpeed TAS |
|  | GlideSpeed IAS | $287 \mathrm{~km} / \mathrm{h}$ |
|  | Sinkspeed True | $217 \mathrm{~km} / \mathrm{h}$ |
|  | Time Glide | $-1.84 \mathrm{~m} / \mathrm{s}$ |
|  | Altitude loss | 21.8 min |
|  | Time Climb | -2410 m |
|  |  | 7.3 min |
| Finalglide | Time not climb | -3.5 min |
|  |  |  |

## Flying an EB29 with 25.3 m span @ $61.7 \mathrm{~kg} / \mathrm{m} 2$

What if we would switch gliders to the single seater EB29, with short wingspan at the higher max take-off weight of 900 kg ?

With a roughly estimated decrease in climb rate of $0.1 \mathrm{~m} / \mathrm{s}$ compared to the lighter EB28 edition, the max average speed in perfect weather would increase to an incredible $258.5 \mathrm{~km} / \mathrm{h}$ !

To reach this speed, however, one would need to fly beyond the red line in sink.

| Taskparameters |  |  |
| :--- | :--- | ---: |
|  | Taskdistance | 101 km |
|  | Total Distance | 104.5 km |
|  | De-tours | $3.50 \%$ |
|  |  |  |
|  | Average Altitude | 4500 m |
|  | Altitude Effect Factor | 1.32 |
|  | Start-Finish Alt Difference | 1150 m |
|  | Climbrate | $5.40 \mathrm{~m} / \mathrm{s}$ |
|  |  | $0.75 \mathrm{~m} / \mathrm{s}$ |
|  | Netto cloud streeting climb |  |


| Output |  |  |
| :--- | :--- | :---: |
|  | timeclimb | 3.7 min |
|  | Altclimb | 1197 m |
|  | timeglide | 19.8 min |
|  | LD | 45 |
|  |  |  |
|  | Total time | $\mathbf{2 3 : 2 7} \mathbf{~ m i n : s e c}$ |
|  | Average Speed | $\mathbf{2 5 8 . 5} \mathbf{~ k m} / \mathbf{h}$ |
|  |  |  |


| Calculations |  |  |
| :--- | :--- | ---: |
| Thermalling | Average netto climb | $6.30 \mathrm{~m} / \mathrm{s}$ |
|  |  |  |
|  | Glide | $318 \mathrm{~km} / \mathrm{h}$ |
|  | GlideSpeed IAS | $240 \mathrm{~km} / \mathrm{h}$ |
|  | Sinkspeed True | $-1.98 \mathrm{~m} / \mathrm{s}$ |
|  | Time Glide | 19.8 min |
|  | Altitude loss | -2347 m |
|  | Time Climb | 7.2 min |
|  |  |  |
|  | Time not climb | -3.5 min |
|  |  |  |

## Flying an ASG29-18m @ 57.1kg/m2

Since the glide polars 18 m and open class ships cross at high speeds, the question is often asked: when would a 18 m glider become better than the open class ships? Would an 18 m glider be more suited for this 100 km speed triangle? According to this model, the answer is: no.

Even if we force the climb rate in the thermal to be the same as for the EB28 edition ( $5.50 \mathrm{~m} / \mathrm{s}$ ), the ASG2918 m at max. weight would be $10 \mathrm{~km} / \mathrm{h}$ slower. In reality, the 18 meter glider would climb a bit worse.

On the other hand, these speed polars are not perfect in this very high speed range. So an exact conclusion is not possible. I have no doubt, however, that an EB29-25.3m at 900kg will always be faster than any current 18 m glider.

| Taskparameters |  |  |
| :--- | :--- | ---: |
|  | Taskdistance | 101 km |
|  | Total Distance | 104.5 km |
|  | De-tours | $3.50 \%$ |
|  |  |  |
|  | Average Altitude | 4500 m |
|  | Altitude Effect Factor | 1.32 |
|  | Start-Finish Alt Difference | 1150 m |
|  | Climbrate | $5.50 \mathrm{~m} / \mathrm{s}$ |
|  |  |  |
|  | Netto cloud streeting climb | $0.75 \mathrm{~m} / \mathrm{s}$ |


| Output |  |  |
| :--- | :--- | :---: |
|  | timeclimb | 4.0 min |
|  | Altclimb | 1330 m |
|  | timeglide | 22.7 min |
|  | LD | 42 |
|  |  |  |
|  | Total time | $\mathbf{2 6 : 4 3} \mathbf{~ m i n : s e c}$ |
|  | Average Speed | $\mathbf{2 2 6 . 8} \mathbf{~ k m} / \mathbf{h}$ |
|  |  |  |


| Calculations |  |  |
| :--- | :--- | ---: |
| Thermalling | Average netto climb | $6.74 \mathrm{~m} / \mathrm{s}$ |
|  | Glide | GlideSpeed TAS |
|  | GlideSpeed IAS | $276 \mathrm{~km} / \mathrm{h}$ |
|  | Sinkspeed True | $209 \mathrm{~km} / \mathrm{h}$ |
|  | Time Glide | $-1.82 \mathrm{~m} / \mathrm{s}$ |
|  | Altitude loss | 22.7 min |
|  | Time Climb | -2480 m |
|  |  | 7.5 min |
| Finalglide | Time not climb | -3.5 min |

## The perfect flight in practice

## Task setting

You have to plan your task, taking the weather into account. Very often we are too lazy to make new waypoints for a single flight.

For the 100 km triangle, this is not a good way to go. As shown previously, making a task exactly 101 km long with 2 new turnpoints, instead of using 2 existing TPs with, for example, a 112 km task length, makes a 5 kph difference, without having to fly better.

If you know that cloudstreets are forecasted in one direction, you should make a one leg as long as possible parallel to that direction!

Within the limit of the rules, the longest leg can be max. $44 \%$ of the total task length. The other two legs, which will cross the cloudstreets under an angle, will both be 28 km long.

## Wind

To keep things simple, I ignore windspeeds for all these calculations. I don't think this record can be flown in very high windspeeds anyway. However, one important remark: set your task so, that your thermaling is all done on a tailwind leg.

## 1 or several thermals?

There are several arguments for using only 1 thermal:

- You only need to center the thermal once, and stay in that thermal for a long climb. The centering part will thus be relatively lower, thus increasing the average climb rate.
- It is pretty much impossible to find 2 equally good perfect thermals in one 100 km flight. One will always be better, and you should thus climb your needed altitude in that thermal.
- Using only 1 thermal, you reduce the amount of dynamic transitions (pulling and pushing), resulting in less losses.

However, there is one good argument for using 2 (and perhaps 3 thermals): using multiple thermals, you will fly on average higher than using only one thermal, thus profiting more from the altitude effect.

The following figure shows the schematic barographs of pilot $A$, who uses two thermals, and stays on average during his flight at 4625 m. Pilot B only uses only the second thermal, and flies on average in 4500m.

The net effect of this would be an increase from $214.9 \mathrm{~km} / \mathrm{h}$ average speed for Pilot B to $216 \mathrm{~km} / \mathrm{h}$ for Pilot B.

The positive effect of using multiple thermals is thus rather small (ca $1 \mathrm{~km} / \mathrm{h}$ ), and I think the net effect of disadvantages of using multiple thermals is larger than the advantage.


## Thermal at begin or end of the course?

So, now that we decided to use only one thermal, where should that thermal be taken?
This can be done at the beginning, middle, or end.
If one would take the thermal right at the end, you would be able to very precisely climb up till the 1000 m altitude difference limit, and thus avoid any waste. This is optimal, and cannot be done when thermaling in the beginning of the course, as you don't know yet which netto climbs and sinks you will encounter during your glide, and thus not know your glide ratio.

Additionally, when you take the thermal at the beginning, you need to start much lower than the cloudbase to leave room for your climb, and thus your flight will be on average flown lower with the same
amount, and thus reduces the altitude effect. This disadvantage is very significant. In the case of starting 500 m lower, the max record speed is lowered from $214.9 \mathrm{~km} / \mathrm{h}$ to $210.8 \mathrm{~km} / \mathrm{h}$. You thus give $4 \mathrm{~km} / \mathrm{h}$ away!

However, there is a still a very good reason to take the thermal in the beginning. If you would take the thermal in the end, you first need to fly the whole course. The chance of finding an excellent climb at the end of your flight, precisely where you need it to be, are very slim. Conversely, if you would wait near the startline until you can easily wait until you have the perfect thermal just in the right location after the startline.

Additionally, a failed attempt with a thermal at the end of the course, will take you 45 min to 1 hour. A failed attempt while taking the thermal in the beginning of the course, might take you only 15 min to 0.5 hour. You could thus make many more attempts in one day when taking your thermal in the beginning.

Conclusion: in theory, taking the climb at the end is much better. In practice, taking the climb in the beginning makes it easier to succeed.

## Final strategy

0 . Wait in front of startline until the perfect climb is on track, close to the startpoint.

1. Dive to the startline at VNE, ca900m below cloudbase, and Pull back gently to STF IAS 210$215 \mathrm{~km} / \mathrm{h}$ in still air.
2. After ca $3-10 \mathrm{~km}$, take climb, and climb ca 1000 m .
3. Fly the rest of the course with MC4 following the energylines, and slowing down to IAS170km/h for taking the 2 turnpoints.
4. Just before crossing the finishline, pull-up to IAS $110 \mathrm{~km} / \mathrm{h}$, and cross 950 m below start altitude, ca 1850 m below cloudbase.

Simple, isn't it? © ;

Note: Off course, as explained before, the optimal strategy is to start at cloudbase, and take your single thermal just before crossing the finish line. However, the chance of success is much lower.

## Flight preparation

## The "Kante"

The Kante is the Namibian ancient and eroded mountain range, which forms the natural barrier between the Namib Desert and the Kalahari Savannnah. Overhead and along this range, the dry Southern Polar airmass, and the wet Upper Tropical airmass will sometimes collide. When this happens, a fantastic convergence line is formed, with the good climbrates on the Eastern side towards the Kalaharo. This phenomenon is then used by happy glider pilots to fly tremendous distances at astonishing speeds.

The ideal location for the 100 km triangle speed run.


## Weather forecast

After one week in Namibia, the topmeteo forecast seemed to us very accurate in that region. For 6 januari 2015, fantastic weather was forecasted in the Windhoek CTR and north of that. Unfortunately the northern half of Namibia is off limits for glider pilots. So only the smallish area from Gollschau to Sosus Valley was predicted to be good.

South of the line Sosus Valley - Mariental, heavy Cirrus clouds would block thermal development. And East of the Kante, only blue thermals were forecasted.



However, in the small area around Gollschau, a prediction of a convergence line was clear from the charts, with a cloudbase up till 5300meter MSL.

As very large flights would be difficult that day, to us, this seemed to be the perfect day to try the 100 km speed triangle.

The optimal time window for the record would be 15.00 to 17.30 local time.
The area of interest is shown with the black circle in the map.


Also windspeeds at altitude were negligible, and mostly coming from the north. In the west of the task area, the wind would turn to the east perpendicular to the convergence line, and speeds would pick up.


## Task setting

The task was chosen to be a bit away from the "Kante", to avoid the usual overdevelopments.
The TP Gollschau-East was chosen as the starting point and finish point.
As the convergence line in that area is mostly north-south, the large leg was laid along that axis and maximized to $43.5 \mathrm{~km} / \mathrm{h}$. The two other legs would be crossing the convergence line main axis, and thus be the shortest of 28.7 km , for a combined total of 101.0 km . To create this task, two new turnpoints had to be created (which were named TBH16 and TBH17).

The triangle could still be chosen to be flown clockwise or counterclockwise. Since we planned to take the single thermal on the first leg, and since slight northerly wind was predicted. We chose to fly counterclockwise.


## The record flight

## Flying to the startline

The startpoint is 140 km from Kiripotib Flying Lodge. Since there was no use in arriving there before 14.00, we chose to take it easy with a late launch at 12.14 local time.


When we arrived in Gollschau Area around 14.00, the convergence line was still building up in the west. Over the start point the situation was reasonable, but in the southern part of the first leg, and on leg2 and leg3, conditions were still blue, with sometimes a small cumulus cloud. It was still too early. In the north (outside of the Namibian glider zone) the conditions looked fantastic.


## $1^{\text {st }}$ Test: Flying through the start gate

Since we were early, but there, we decided to do an initial run through the startgate to practice the pull up maneuver with the EB28, and check the instruments and Nav box information on the LX9000.

## $2^{\text {nd }}$ attempt: A Belgian record, and encouraging results on the first trial

It was still way too blue and early, but now we wanted to try a full lap, to check every aspect of the flight.
We started with a groundspeed of $305 \mathrm{~km} / \mathrm{h}$ in 4599 m , and pulled under the first cumulus which we believed to have a good thermal after 17 km .

Nothing.
1 useless circle, we pushed on ahead again.
After yet another 11 km , we found a poor $3.3 \mathrm{~m} / \mathrm{s}$ climb. Since this was just a trial we decided to climb in it for 719m.

After leaving, and flying another 10 km , we found another thermal. $4.5 \mathrm{~m} / \mathrm{s}$ this time. We climbed for 236 m up till cloudbase, which was only 4930 m under that cloud.

After taking the turnpoint, it became clear that without a decent energyline, we were again sinking under glide path. We thus turned into the third thermal on the second leg. $4.3 \mathrm{~m} / \mathrm{s}$ over 415 m .

This was more than sufficient to complete the rest of the lap, and we crossed the finish line in 3748 , 851m below the start, and thus no pull-up needed.

Because of the blueish conditions, the cloudstreeting was bad (only $0.21 \mathrm{~m} / \mathrm{s}$ compared to Laszlo's $0.46 \mathrm{~m} / \mathrm{s}$ ), and the glide ratio was thus bad. Because of that, we needed to take 3 thermals. Average climb rate was with $3.78 \mathrm{~m} / \mathrm{s}$ also not good at all (although perhaps it could have been improved climbing more in the $2^{\text {nd }}$ and $3^{\text {rd }}$ thermal instead of the $1^{\text {st }}$ ).

We also thermaled ca 300 m too much.
To top it off, we made a useless circle...
With all this, we still managed to fly $190.03 \mathrm{~km} / \mathrm{h}$.
This was just $2.6 \mathrm{~km} / \mathrm{h}$ below the current record claim, and $20 \mathrm{~km} / \mathrm{h}$ above the standing Belgian record.
Very encouraging indeed.

| Alt. Loss in Glide | -2223 m |
| :--- | :---: |
| Time Glide | 1550 sec |
| Average Sinkrate | $-1.43 \mathrm{~m} / \mathrm{s}$ |
| Average Glidespeed TAS | $242 \mathrm{~km} / \mathrm{h}$ |
| Polar Sinkrate for that TAS | $-1.65 \mathrm{~m} / \mathrm{s}$ |
| Netto Avg. Airmass movement | $\mathbf{0 . 2 1} \mathrm{m} / \mathrm{s}$ |

The actual flight statistics of attempt 2 in the model

| Taskparameters |  |  |
| :--- | :--- | ---: |
|  | Taskdistance | 101 km |
|  | Total Distance | 104.4 km |
|  | De-tours | $3.37 \%$ |
|  |  |  |
|  | Average Altitude | 4250 m |
|  | Altitude Effect Factor | 1.30 |
|  | Start-Finish Alt Difference | 851 m |
|  | Climbrate | $3.78 \mathrm{~m} / \mathrm{s}$ |
|  |  | $0.21 \mathrm{~m} / \mathrm{s}$ |
|  | Netto cloud streeting climb |  |


| Calculations |  |  |
| :--- | :--- | :---: |
| Thermalling | Average netto climb | $4.55 \mathrm{~m} / \mathrm{s}$ |
|  |  |  |
|  | GlideSpeed TAS | $242 \mathrm{~km} / \mathrm{h}$ |
|  | GlideSpeed IAS | $186 \mathrm{~km} / \mathrm{h}$ |
|  | Sinkspeed True | $-1.43 \mathrm{~m} / \mathrm{s}$ |
|  | Time Glide | 25.9 min |
|  | Altitude loss | -2227 m |
|  | Time Climb | 9.8 min |
|  |  |  |
|  | Time not climb | -3.8 min |
|  |  |  |


| Output |  |  |
| :--- | :--- | :---: |
|  | timeclimb | 6.1 min |
|  | Altclimb | 1376 m |
|  | timeglide | 25.9 min |
|  | LD | 47 |
|  |  |  |
|  | Total time | $\mathbf{3 1 : 5 7} \mathbf{~ m i n : s e c}$ |
|  | Average Speed | $\mathbf{1 8 9 . 6} \mathbf{~ k m} / \mathbf{h}$ |
|  |  |  |

In hindsight, if we would have used up the whole 1150 m altitude bonus, skipped the useless turn, thermaled less in the $1^{\text {st }}$, and a bit more in the $2^{\text {nd }}$ or $3^{\text {rd }}$ climb (resulting in a $4.00 \mathrm{~m} / \mathrm{s}$ average climbrate), and fly a bit faster ( $250 \mathrm{~km} / \mathrm{h}$ TAS) we would have been able to break the $200 \mathrm{~km} / \mathrm{h}$ barrier already.

| Taskparameters |  |  |
| :---: | :---: | :---: |
|  | Taskdistance | 101 km |
|  | Total Distance | 104.4 km |
|  | De-tours | 3.37\% |
|  | Average Altitude | 4250 m |
|  | Altitude Effect Factor | 1.30 |
|  | Start-Finish Alt Difference | 1150 m |
|  | Climbrate | $4.00 \mathrm{~m} / \mathrm{s}$ |
|  | Netto cloud streeting climb | $0.21 \mathrm{~m} / \mathrm{s}$ |
| Calculations |  |  |
| Thermalling | Average netto climb | $4.77 \mathrm{~m} / \mathrm{s}$ |
| Glide | GlideSpeed TAS | 250 km/h |
|  | GlideSpeed IAS | 192 km/h |
|  | Sinkspeed True | $-1.59 \mathrm{~m} / \mathrm{s}$ |
|  | Time Glide | 25.1 min |
|  | Altitude loss | -2389 m |
|  | Time Climb | 10.0 min |
| Finalglide | Time not climb | -4.8 min |


| Output |  |  |
| :--- | :--- | :---: |
|  | timeclimb | 5.2 min |
|  | Altclimb | 1239 m |
|  | timeglide | 25.1 min |
|  | LD | 44 |
|  |  |  |
|  | Total time | $\mathbf{3 0 : 1 3} \mathbf{~ m i n}: \mathbf{s e c}$ |
|  | Average Speed | $\mathbf{2 0 0 . 5} \mathbf{~ k m} / \mathbf{h}$ |
|  |  |  |

## Attempts 3\&4

We did a couple of other runs and attempts, but each failed and was thus abandoned for its own reason (low LD, bad centering, no good climb ...).

```
5 th attempt: Success!!!
```

After two hours of playing around, the topmeteo forecast we hoped for started to come true. The convergence line was now really getting strong, but a bit more in the West than we had hoped for. We would need to fly a detour to reach it. The second and third leg were still with separate cumulus clouds, but also these had grown much stronger.

The wind had turned a bit to the west (as forecasted), but we would still tailwind on the first leg.
So, at 16:15 we crossed the startline in 4161 m . We immediately turned to the right, to the convergence line in the west, and we reached it after 15 km . Not far under the massive cloudstreet, an authorative push under the left wing signaled a thermal. This would turn out to be a $4.36 \mathrm{~m} / \mathrm{s}$ thermal. Not the best ever, but quite good nonetheless. We climbed for 1069 meters, and left a bit below cloudbase.

The powerful convergence line was outlined a few degrees off our track, but the airmass moved us clearly upward underneath it.

A few km before the first turnpoint (TBH16), we said goodbye to the convergence line, and focused on taking the corner.

The second leg was like in the second attempt of the day. But with the tremendous amount of altitude left, we could push on to the third turnpoint.

While taking the last corner of the day, we knew this should suffice. 900 m left for 28 km , and speed to pull up another 200 m . On top of that, a large and promising cumulus was just building up, a bit southwest of track, just a few kilometer before the finish line.

After the final dash we stopped the clock:

29:26
$205.89 \mathrm{~km} / \mathrm{h}$ over the 101 km task.

Adrenaline still pumping, and a battle cry of joy. This speed run truly is fun!
A few moments after crossing the finish line, we enter yet another great thermal (even better than the one we had on the record lap). I take a picture of the LX9000 and post it on facebook.

We decide that we worked enough for today, and unhurriedly fly back to Kiripotib.

Sattelite image of the clouds during the successful speed run


Our Diamond shaped trackline





## Post Flight Analysis

We can see that the Netto Avg. Airmass movement is quite good thanks to the massive convergence line. It is ca $0.1 \mathrm{~m} / \mathrm{s}$ better than during Laszlo's record attempt. Climbrate in the thermal is however worse $(4.36 \mathrm{~m} / \mathrm{s})$.

Detours where average, but could be smaller.
This was definitely good weather, but not exceptional.

| Alt. Loss in Glide | -1819 m |
| :--- | :---: |
| Time Glide | 1521 sec |
| Average Sinkrate | $-1.20 \mathrm{~m} / \mathrm{s}$ |
| Average Glidespeed TAS | $250 \mathrm{~km} / \mathrm{h}$ |
| Polar Sinkrate for that TAS | $-1.75 \mathrm{~m} / \mathrm{s}$ |
| Netto Avg. Airmass movement | $\mathbf{0 . 5 5} \mathrm{m} / \mathrm{s}$ |


| Taskparameters |  |  |
| :--- | :--- | :---: |
|  | Taskdistance | 101 km |
|  | Total Distance | 105.6 km |
|  | De-tours | $4.55 \%$ |
|  |  |  |
|  | Average Altitude | 4500 m |
|  | Altitude Effect Factor | 1.32 |
|  | Start-Finish Alt Difference | 749 m |
|  | Climbrate | $4.36 \mathrm{~m} / \mathrm{s}$ |
|  |  |  |
|  | Netto cloud streeting climb | $0.55 \mathrm{~m} / \mathrm{s}$ |
|  |  |  |


| Output |  |  |
| :--- | :--- | :---: |
|  | timeclimb | 4.1 min |
|  | Altclimb | 1077 m |
|  | timeglide | 25.3 min |
|  | LD | 58 |
|  |  |  |
|  | Total time | $29: 27 \mathrm{~min}: \mathbf{s e c}$ |
|  | Average Speed | $205.7 \mathrm{~km} / \mathrm{h}$ |


| Calculations |  |  |
| :--- | :--- | :---: |
| Thermalling | Average netto climb | $5.16 \mathrm{~m} / \mathrm{s}$ |
|  |  |  |
|  | Glide | GlideSpeed TAS |
|  | Sinkspeed IAS | $250 \mathrm{~km} / \mathrm{h}$ |
|  | Time Glide | $189 \mathrm{~km} / \mathrm{h}$ |
|  | Altitude loss | $-1.20 \mathrm{~m} / \mathrm{s}$ |
|  | Time Climb | 25.3 min |
|  |  | -1826 m |
|  | Time not climb | 7.0 min |
|  |  | -2.9 min |

Our glidespeed was only $12 \mathrm{~km} / \mathrm{h}$ IAS under the optimal STF $267 \mathrm{~km} / \mathrm{h}$ TAS. The resulting effect of doing it perfectly would only have been $1 \mathrm{~km} / \mathrm{h}$ more on the average speed.

| Taskparameters |  |  |
| :--- | :--- | :---: |
|  | Taskdistance | 101 km |
|  | Total Distance | 105.6 km |
|  | De-tours | $4.55 \%$ |
|  | Average Altitude | 4500 m |
|  | Altitude Effect Factor | 1.32 |
|  | Start-Finish Alt Difference | 749 m |
|  | Climbrate | $4.36 \mathrm{~m} / \mathrm{s}$ |
|  |  |  |
|  | Netto cloud streeting climb | $0.55 \mathrm{~m} / \mathrm{s}$ |
|  |  |  |


| Output |  |  |
| :--- | :--- | :---: |
|  | timeclimb | 5.6 min |
|  | Altclimb | 1460 m |
|  | timeglide | 23.7 min |
|  | LD | 48 |
|  |  |  |
|  | Total time | $\mathbf{2 9 : 1 9} \mathbf{~ m i n} \mathbf{s e c}$ |
|  | Average Speed | $\mathbf{2 0 6 . 7} \mathbf{~ k m} / \mathbf{h}$ |
|  |  |  |


| Calculations |  |  |
| :--- | :--- | :---: |
| Thermalling | Average netto climb | $5.16 \mathrm{~m} / \mathrm{s}$ |
|  | Glide | GlideSpeed TAS |
|  | GlideSpeed IAS | $267 \mathrm{~km} / \mathrm{h}$ |
|  | Sinkspeed True | $201 \mathrm{~km} / \mathrm{h}$ |
|  | Time Glide | $-1.55 \mathrm{~m} / \mathrm{s}$ |
|  | Altitude loss | 23.7 min |
|  | Time Climb | -2209 m |
|  |  | 8.4 min |
| Finalglide | Time not climb | -2.9 min |
|  |  |  |

However, we only used up 749 m of our 1150 m bonus. We could have thermaled 390 m less, and then $218.1 \mathrm{~km} / \mathrm{h}$ would have been reached! However, again, this is hindsight knowledge. The glide ratio could have been much worse.

| Calculations |  |  |
| :--- | :--- | :---: |
| Thermalling | Average netto climb | $5.16 \mathrm{~m} / \mathrm{s}$ |
| Glide | GlideSpeed TAS | $267 \mathrm{~km} / \mathrm{h}$ |
|  | GlideSpeed IAS | $201 \mathrm{~km} / \mathrm{h}$ |
|  | Sinkspeed True | $-1.55 \mathrm{~m} / \mathrm{s}$ |
|  | Time Glide | 23.7 min |
|  | Altitude loss | -2209 m |
|  | Time Climb | 8.4 min |
|  |  |  |
| Finalglide | Time not climb | -4.4 min |
|  |  |  |


| Output |  |  |
| :--- | :--- | :---: |
|  | timeclimb | 4.0 min |
|  | Altclimb | 1059 m |
|  | timeglide | 23.7 min |
|  | LD | 48 |
|  |  |  |
|  | Total time | $\mathbf{2 7 : 4 7} \mathbf{~ m i n : s e c}$ |
|  | Average Speed | $\mathbf{2 1 8 . 1} \mathbf{~ k m} / \mathbf{h}$ |

An interesting thought experiment again: what would have been our average speed, if there would not have been allowed a difference between start and finish altitude. It would thus negate the "bonus". This is called the XC speed in the LX9000 and Seeyou software.

For this record lap, the XC speed is $187.5 \mathrm{~km} / \mathrm{h}$. This is very high for a set task, and something you would rarely see in a normal flight!

| Taskparameters |  |  |
| :--- | :--- | ---: |
|  | Taskdistance | 101 km |
|  | Total Distance | 105.6 km |
|  | De-tours | $4.55 \%$ |
|  |  |  |
|  | Average Altitude | 4500 m |
|  | Altitude Effect Factor | 1.32 |
|  | Start-Finish Alt Difference | 0 m |
|  | Climbrate | $4.36 \mathrm{~m} / \mathrm{s}$ |
|  |  | $0.55 \mathrm{~m} / \mathrm{s}$ |
|  | Netto cloud streeting climb |  |


| Output |  |  |
| :--- | :--- | :---: |
|  | timeclimb | 7.0 min |
|  | Altclimb | 1826 m |
|  | timeglide | 25.3 min |
|  | LD | 58 |
|  |  |  |
|  | Total time | $\mathbf{3 2 : 1 9 ~ m i n : s e c}$ |
|  | Average Speed |  |


| Calculations |  |  |
| :--- | :--- | :---: |
| Thermalling | Average netto climb | $5.16 \mathrm{~m} / \mathrm{s}$ |
| Glide | GlideSpeed TAS | $250 \mathrm{~km} / \mathrm{h}$ |
|  | GlideSpeed IAS | $189 \mathrm{~km} / \mathrm{h}$ |
|  | Sinkspeed True | $-1.20 \mathrm{~m} / \mathrm{s}$ |
|  | Time Glide | 25.3 min |
|  | Altitude loss | -1826 m |
|  | Time Climb | 7.0 min |
|  | Time not climb | 0.0 min |

## $205.89 \mathrm{~km} / \mathrm{h}$ is not the end

As with previous projects, I always enjoy thinking something up behind the desk, and then testing it out in the real world. I am glad we succeeded, and also glad to be able to contribute to the theoretical understanding of gliding, in this case of the 100 km speed run.

The record we set was decent, but definitely not the end. I predict that before the end of the 2015/2016 season of Namibia, the record speed on the 100 km FAI triangle will be close to 220 kph .

It just takes one man in the right spot with a good glider, and with half an hour time to spare.

## Thanks

We would like to thank:
Bernd Dolba \& Heribert, as well as the whole team of Kiripotib Gliding Lodge for helping and hosting http://www.dolba.de/kiripotib flying/index.html

Klaus Seemann \& Bostjan Pristavec for trusting us with their fantastic glider
Topmeteo for again a perfect forecast
http://de.africa.topmeteo.eu/go/home
Sponsor Brillen Kueppers for the superb new sunglasses
http://www.brillen-kueppers.de/de
Naviter/Seeyou for making flight planning a lot easier
http://www.naviter.com/
LXNav for making flying a lot easier
http://www.lxnav.com/

## Previous reports

Keiheuvel - Viersen - Spanish border: 1000km straight distance
http://www.scribd.com/doc/90226458/16042012-Keiheuvel-Viersen-SpanishBorder

The Cloud Making Machines: 1000km from Flanders
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## Solving the Speed Run

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Tijl Schmelzer \& Bert Sen. Schmelzer

